

15th German LS-DYNA Forum, Bamberg, 2018

# A Hosford-based orthotropic plasticity model in LS-DYNA

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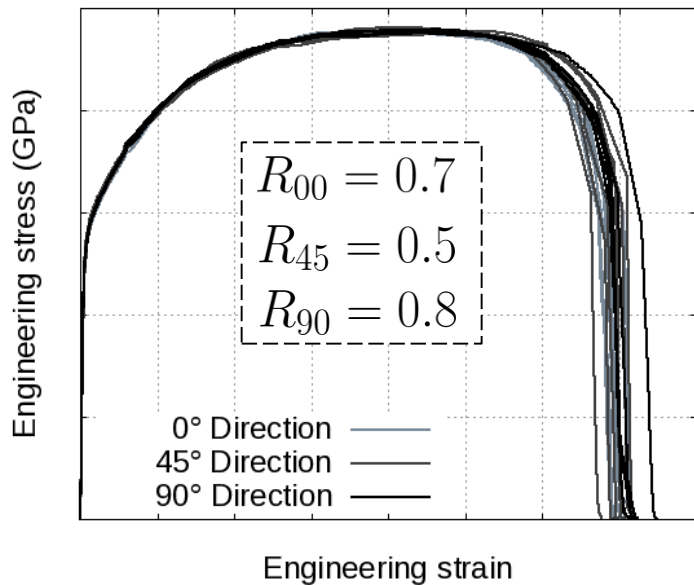
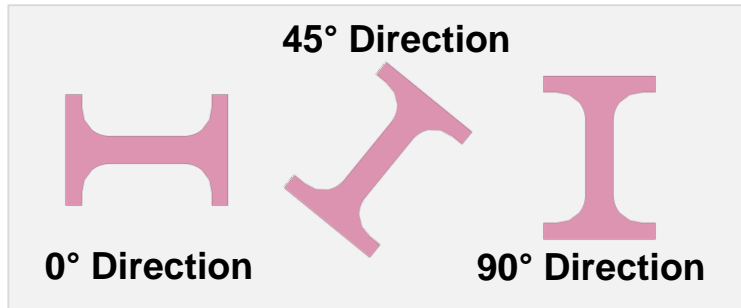
<sup>3</sup>Consultant

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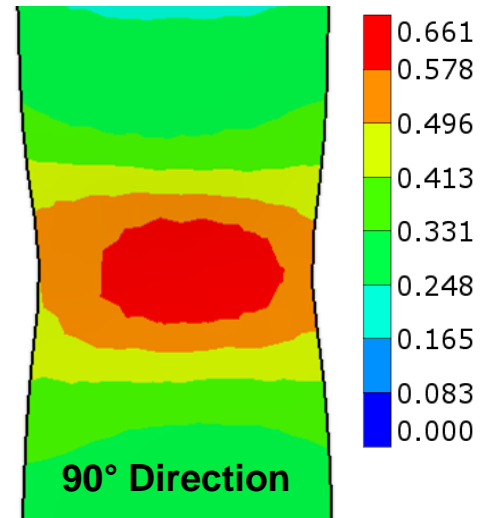
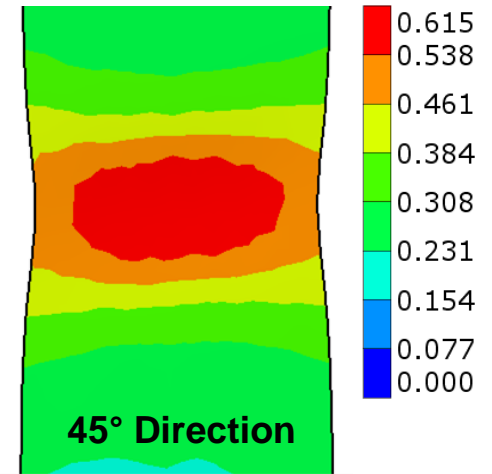
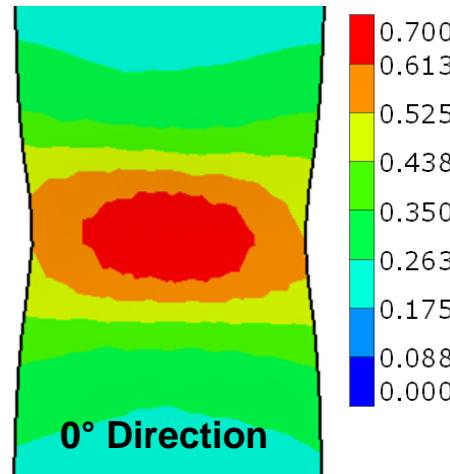
**October 17<sup>th</sup> 2018**

# Orthotropic behavior

Example: aluminum sheet



Strain fields prior to failure (DIC measurement)

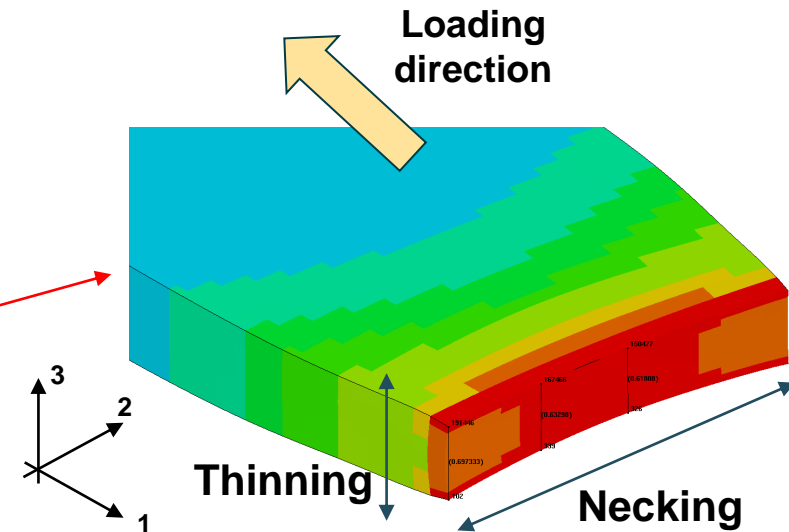
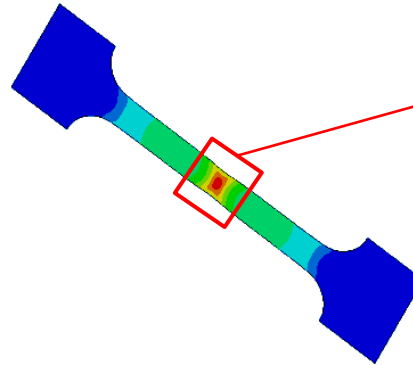


# Orthotropic behavior

The Lankford parameter (R value)

- Definition (uniaxial tension):

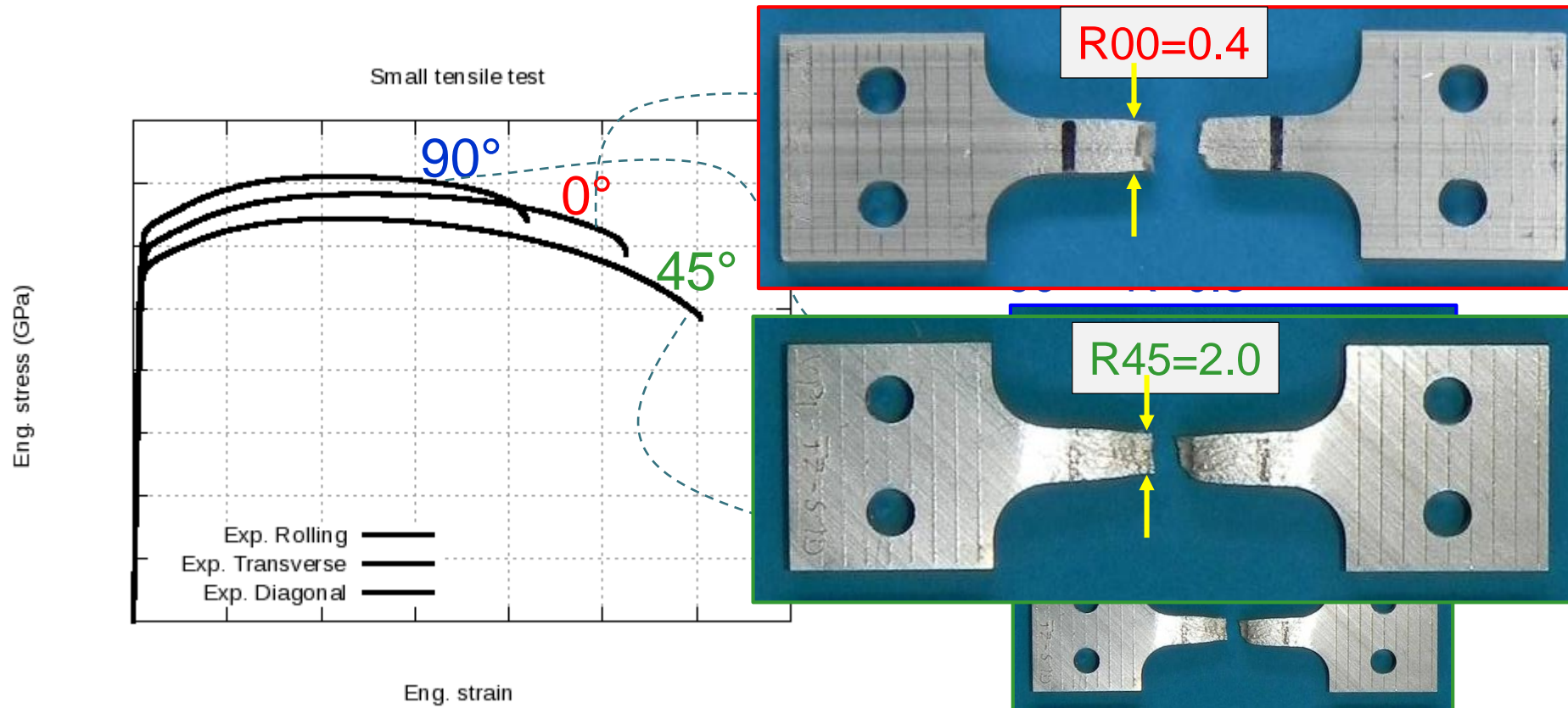
$$R = \frac{\dot{\epsilon}_{22}^p}{\dot{\epsilon}_{33}^p}$$



$R = \frac{\dot{\epsilon}_{22}^p}{\dot{\epsilon}_{33}^p} = 1.0$	$\longrightarrow$	$\dot{\epsilon}_{22}^p = \dot{\epsilon}_{33}^p$	$\longrightarrow$	<b>Thinning and necking are comparable</b>	
$R = \frac{\dot{\epsilon}_{22}^p}{\dot{\epsilon}_{33}^p} < 1.0$	$\longrightarrow$	$\dot{\epsilon}_{22}^p < \dot{\epsilon}_{33}^p$	$\longrightarrow$	<b>More thinning</b>	<b>Less necking</b>
$R = \frac{\dot{\epsilon}_{22}^p}{\dot{\epsilon}_{33}^p} > 1.0$	$\longrightarrow$	$\dot{\epsilon}_{22}^p > \dot{\epsilon}_{33}^p$	$\longrightarrow$	<b>Less thinning</b>	<b>More necking</b>

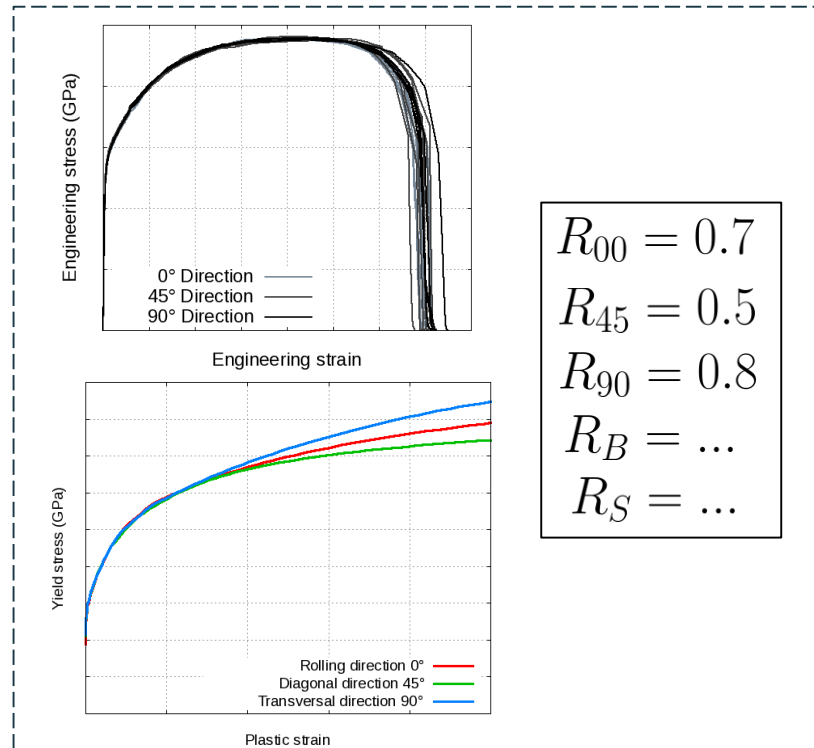
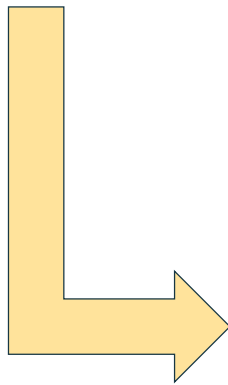
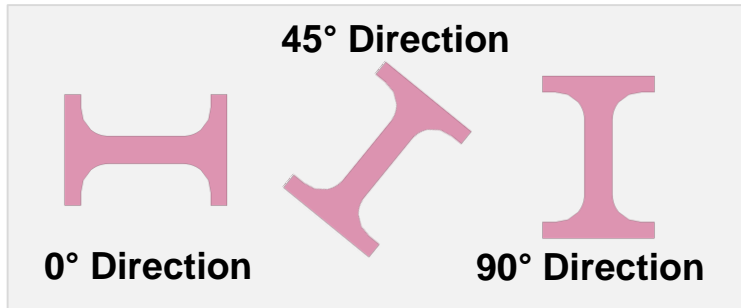
# Orthotropic behavior

Example: aluminum extrusion

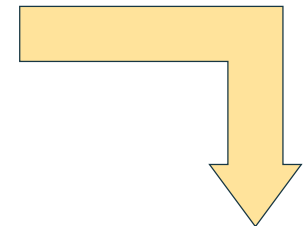


# Orthotropic modeling

From experimental data to simulation parameters



$$\begin{aligned} R_{00} &= 0.7 \\ R_{45} &= 0.5 \\ R_{90} &= 0.8 \\ R_B &= \dots \\ R_S &= \dots \end{aligned}$$



**LS-DYNA**

# MATERIAL MODELING IN LS-DYNA

# \*MAT\_036 + HR=3

Barlat & Lian (1989)

R values are similar

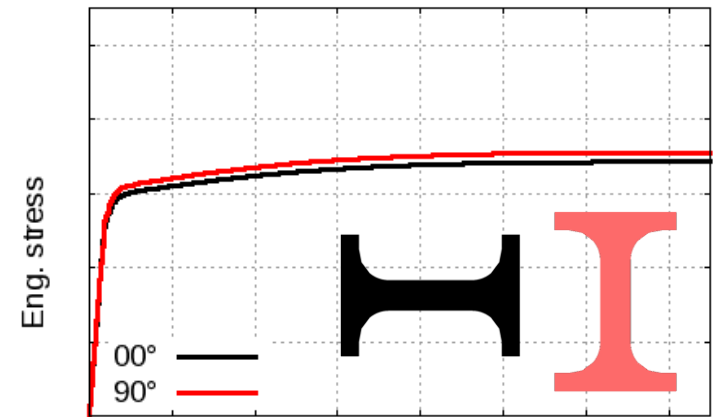
*MAT_3-PARAMETER_BARLAT									
\$	MID	RO	E	PR	HR	P1	P2	ITER	
	1	2.70E-06	70.0	0.3	3				
\$	M	R00	R45	R90	LCID	E0	SPI	P3	
	8.0	0.8	1.0	0.9	100				
\$...									

$$\Phi(\boldsymbol{\sigma}) = \frac{1}{2} (a |K_1 + K_2|^m + a |K_1 - K_2|^m + c |2K_2|^m) - \sigma_y^m = 0$$

$R_{00} = 0.8$   
 $R_{45} = 1.0$   
 $R_{90} = 0.9$   
 $\sigma_y, m$

internal fitting

$a = \dots$   
 $c = \dots$   
 $h = \dots$   
 $p = \dots$



Eng. stress

# \*MAT\_036 + HR=3

Barlat & Lian (1989)

R values differ significantly from each other

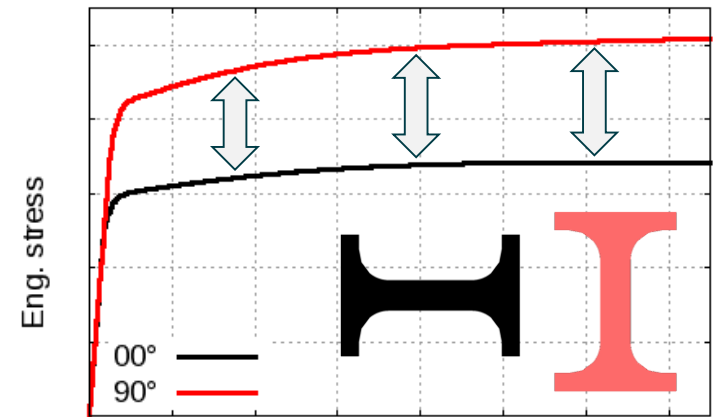
*MAT_3-PARAMETER_BARLAT									
\$	MID	RO	E	PR	HR	P1	P2	ITER	
	1	2.70E-06	70.0	0.3	3				
\$	M	R00	R45	R90	LCID	E0	SPI	P3	
	8.0	0.5	1.0	2.0	100				
\$...									

$$\Phi(\boldsymbol{\sigma}) = \frac{1}{2} (a |K_1 + K_2|^m + a |K_1 - K_2|^m + c |2K_2|^m) - \sigma_y^m = 0$$

$R_{00} = 0.5$   
 $R_{45} = 1.0$   
 $R_{90} = 2.0$   
 $\sigma_y, m$

internal fitting

$a = \dots$   
 $c = \dots$   
 $h = \dots$   
 $p = \dots$



Eng. strain



# \*MAT\_036 + HR=7

Fleischer & Borrvall (2007): modified Barlat & Lian

R values differ significantly from each other

$$R = \frac{\dot{\epsilon}_{22}^p}{\dot{\epsilon}_{33}^p}$$

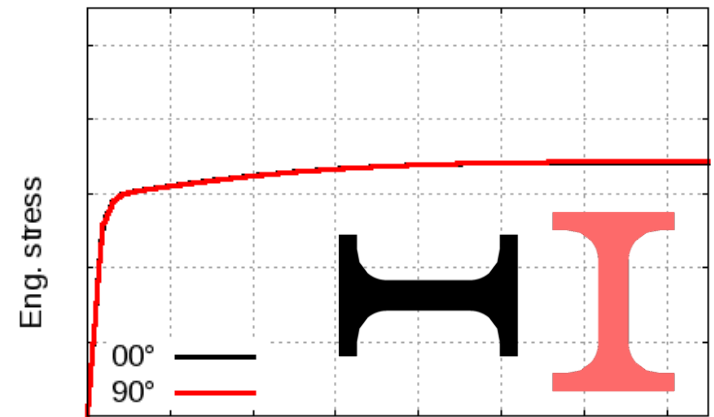
*MAT_3-PARAMETER_BARLAT								
\$	MID	RO	E	PR	HR	P1	P2	ITER
	1	2.70E-06	70.0	0.3	7	145	190	
\$	M	R00	R45	R90	LCID	E0	SPI	P3
	8.0	0.5	1.0	2.0	100			
\$...								

$$\Phi(\boldsymbol{\sigma}) = \frac{1}{2} (a |K_1 + K_2|^m + a |K_1 - K_2|^m + c |2K_2|^m) - \sigma_y^m = 0$$

$R_{00} = 0.5$   
 $R_{45} = 1.0$   
 $R_{90} = 2.0$   
 $\sigma_y, m$

internal fitting

$a = \dots$   
 $c = \dots$   
 $h = \dots$   
 $p = \dots$



Eng. strain

# \*MAT\_036 + HR=7

Fleischer & Borrvall (2007): modified Barlat & Lian

## Yield function

$$\Phi(\boldsymbol{\sigma}) = \frac{1}{2} (a |K_1 + K_2|^m + a |K_1 - K_2|^m + c |2K_2|^m) - \sigma_y^m = 0$$

$$\sigma_y(\boldsymbol{\sigma}, \varepsilon^p) = \alpha_{00} \sigma_y^{00}(\varepsilon^p) + \alpha_{45} \sigma_y^{45}(\varepsilon^p) + \alpha_{90} \sigma_y^{90}(\varepsilon^p) + \alpha_B \sigma_y^B(\varepsilon^p) + \alpha_{shear} \sigma_y^{shear}(\varepsilon^p)$$

## Flow rule

$$\dot{\boldsymbol{\varepsilon}}^p = \begin{bmatrix} \dot{\varepsilon}_{11}^p & \dot{\varepsilon}_{12}^p & \dot{\varepsilon}_{13}^p \\ \dot{\varepsilon}_{21}^p & \dot{\varepsilon}_{22}^p & \dot{\varepsilon}_{23}^p \\ \dot{\varepsilon}_{31}^p & \dot{\varepsilon}_{32}^p & \dot{\varepsilon}_{33}^p \end{bmatrix} = \dot{\gamma} \left( \frac{\partial \Phi}{\partial \boldsymbol{\sigma}} + \Delta \mathbf{N} \right)$$

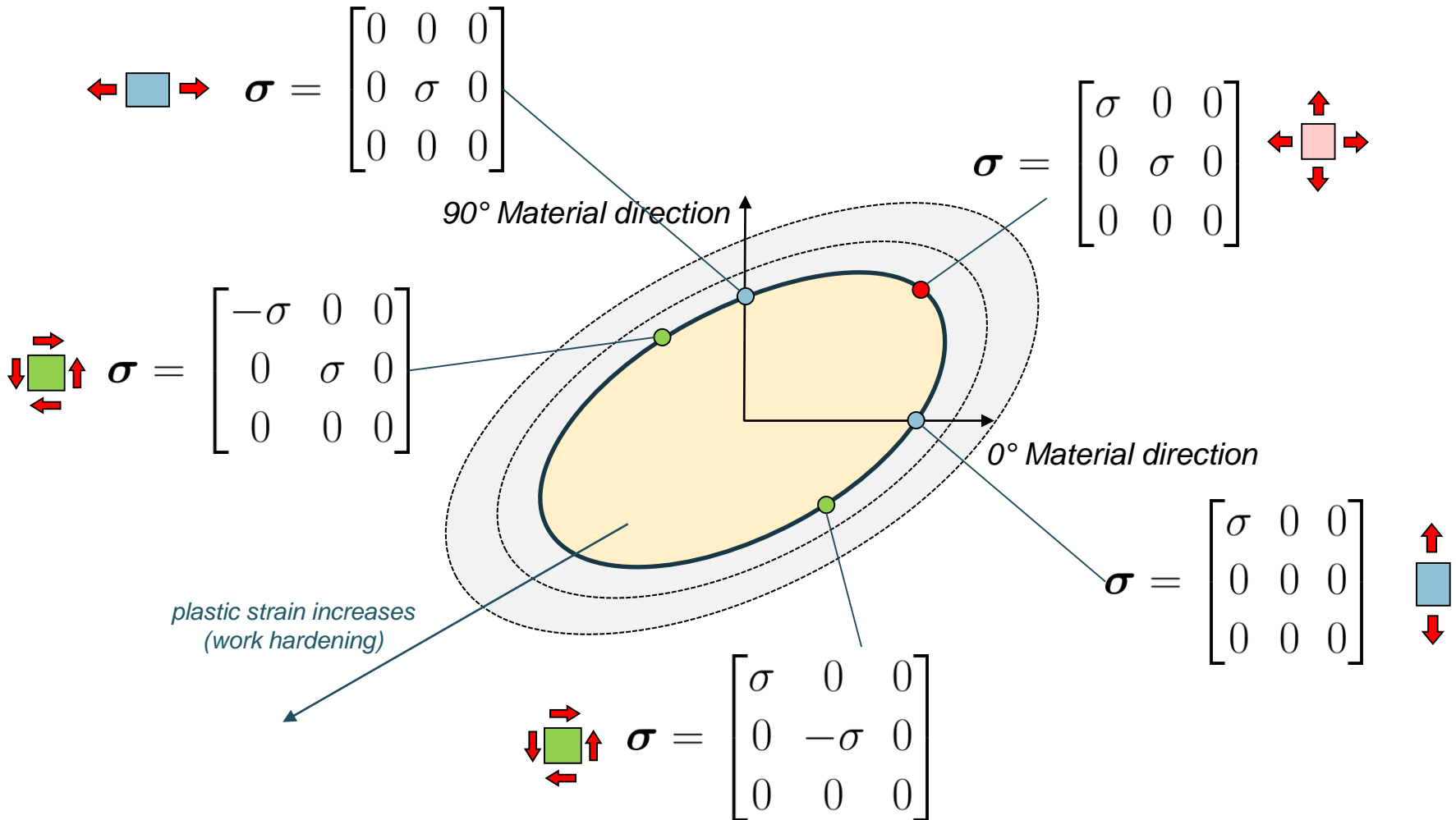
$R = \frac{\dot{\varepsilon}_{22}^p}{\dot{\varepsilon}_{33}^p}$

associated plasticity      smallest necessary increment to match experimental R values

*non-associated plasticity*

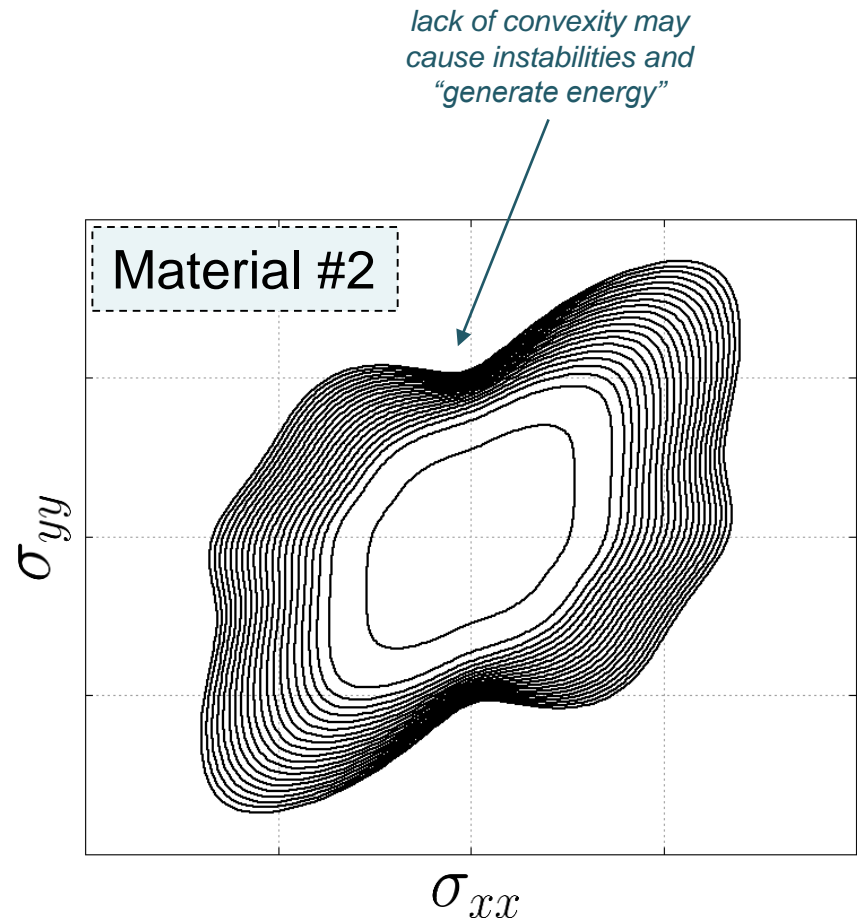
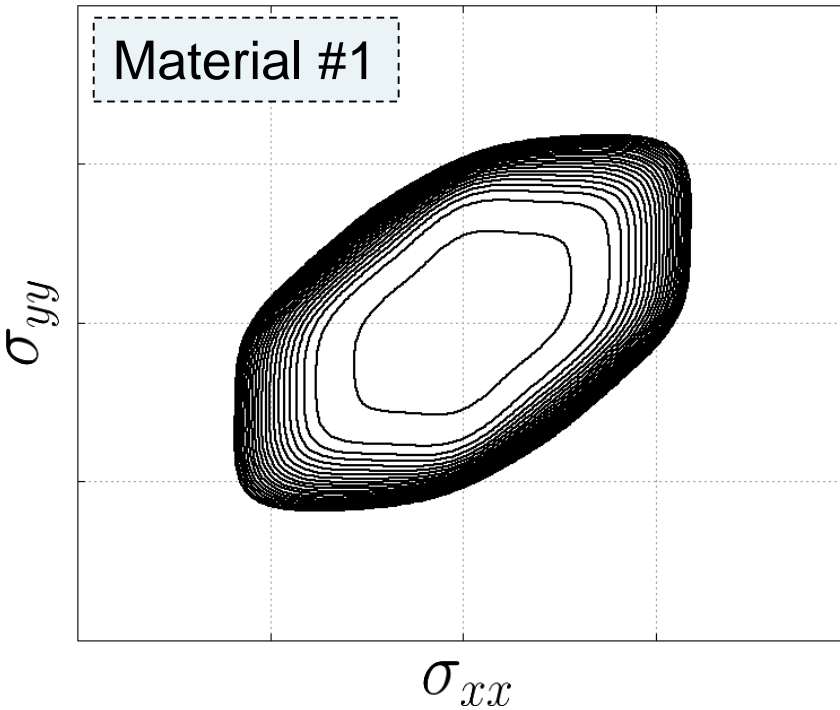
# Yield surface (von Mises)

Example: Yielding at different stress states



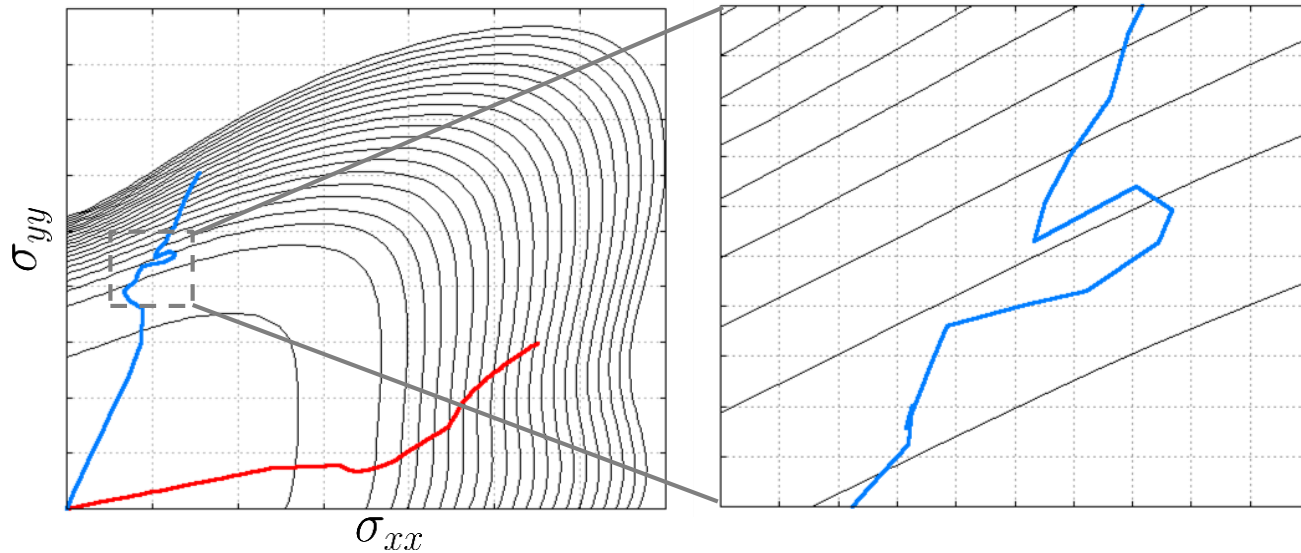
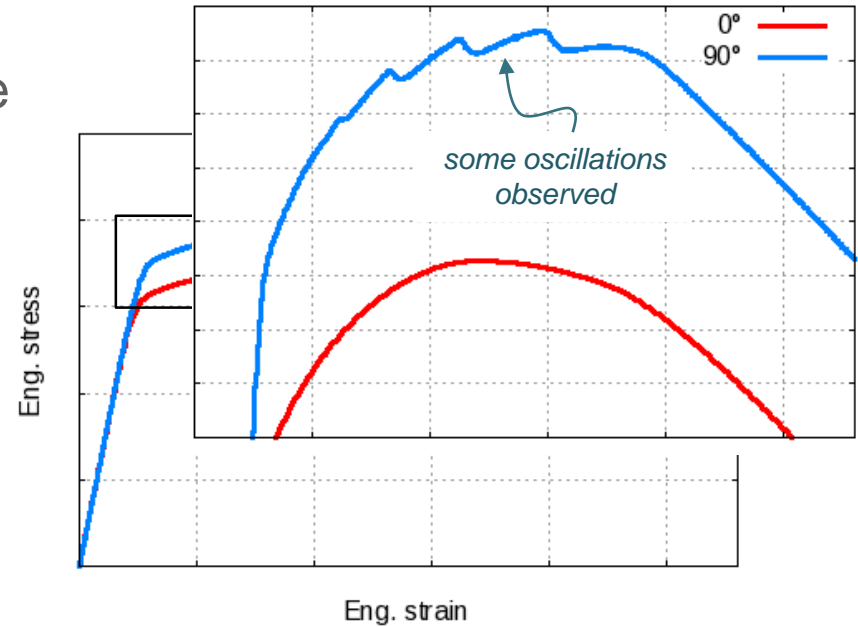
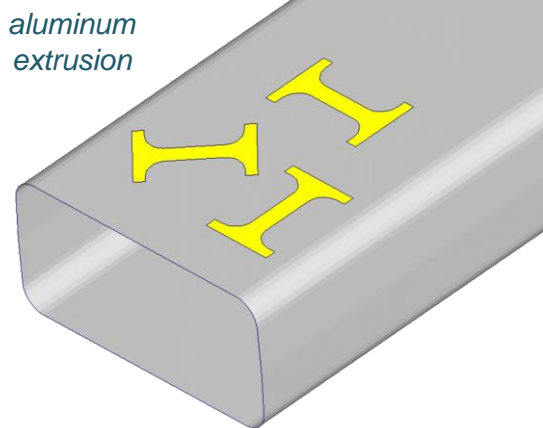
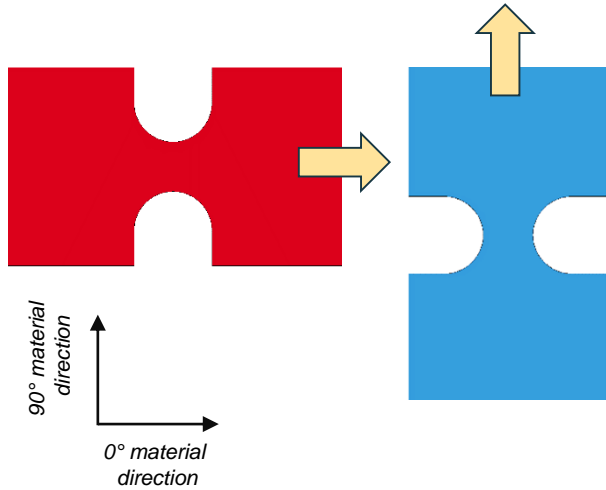
# \*MAT\_036 + HR=7

Yield surface with increasing plastic strain



# \*MAT\_036 + HR=7

Possible effect of a concave yield surface  
Simulation of a notched tensile test



# \*MAT\_036E + HOSF=0/1

Barlat- and Hosford-based orthotropic models

flow rule

$$\dot{\epsilon}^p = \dot{\gamma} \left( \frac{\partial \Phi}{\partial \sigma} + \Delta \mathbf{N} \right)$$

```

*MAT_EXTENDED_3-PARAMETER_BARLAT / *MAT_036E
$      MID      RO      E      PR
      1      2.70E-6      70.0      0.3
$      LCH00      LCH45      LCH90      LCHBI      LCHSH      HOSF
      100      145      190
$      LCR00      LCR45      LCR90      LCRBI      LCRSH      M
      -0.5      -1.0      -2.0
$...
    
```

**HOSF=0**  
(\*MAT\_036+HR=7)

$$\Phi(\sigma) = \frac{1}{2} (a |K_1 + K_2|^m + a |K_1 - K_2|^m + c |2K_2|^m) - \sigma_y^m = 0$$

**HOSF=1**

$$\Phi(\sigma) = \frac{1}{2} (|\sigma_1|^m + |\sigma_2|^m + |\sigma_1 - \sigma_2|^m) - \sigma_y^m = 0$$

No use of a, c, p and h in the yield function  
(i.e., no influence of the R values on the yield function)

# \*MAT\_036E + HOSF=1

Hosford-based orthotropic model

Yielding (hardening curve) and plastic flow  
(R values) are treated separately!

```

*MAT_EXTENDED_3-PARAMETER_BARLAT / *MAT_036E
$      MID      RO      E      PR
      1  2.70E-6  70.0   0.3
$      LCH00    LCH45    LCH90    LCHBI    LCHSH    HOSF
      100      145      190
$      LCR00    LCR45    LCR90    LCRBI    LCRSH
      -0.5     -1.0     -2.0
$...
    
```

Yield function

Flow rule

$$\Phi(\boldsymbol{\sigma}) = \sigma_{eff} - \sigma_y^m = 0$$

$$\sigma_{eff} = \frac{1}{2} (|\sigma_1|^m + |\sigma_2|^m + |\sigma_1 - \sigma_2|^m)$$

$$\dot{\epsilon}^p = \dot{\gamma} \left( \frac{\partial \Phi}{\partial \boldsymbol{\sigma}} + \Delta \mathbf{N} \right)$$

*non-associated plasticity!*

# \*MAT\_036E + HOSF=1

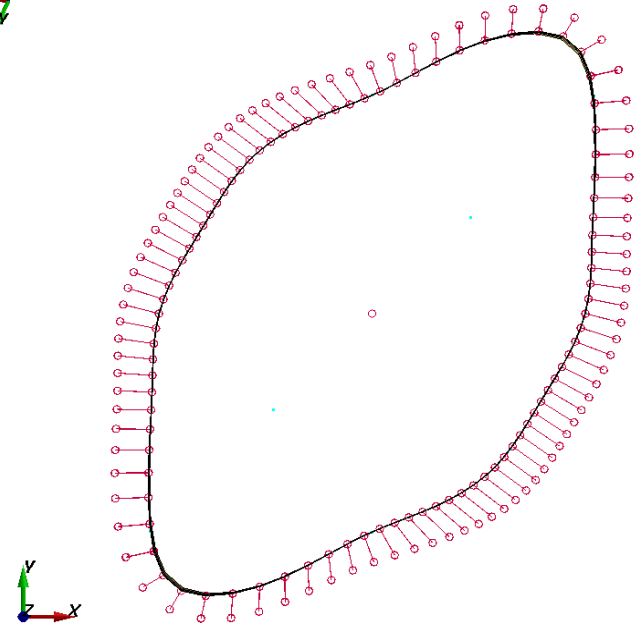
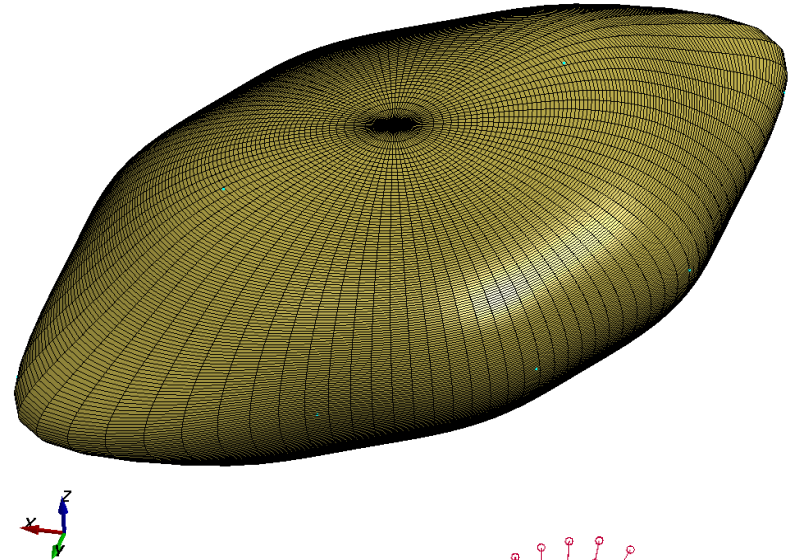
Hosford-based orthotropic model

$$\boldsymbol{\sigma} = \begin{bmatrix} \sigma_{11} & \sigma_{12} & 0 \\ 0 & \sigma_{22} & 0 \\ 0 & 0 & 0 \end{bmatrix}$$

Flow rule

$$\dot{\boldsymbol{\epsilon}}^p = \dot{\gamma} \left( \frac{\partial \Phi}{\partial \boldsymbol{\sigma}} + \Delta \mathbf{N} \right)$$

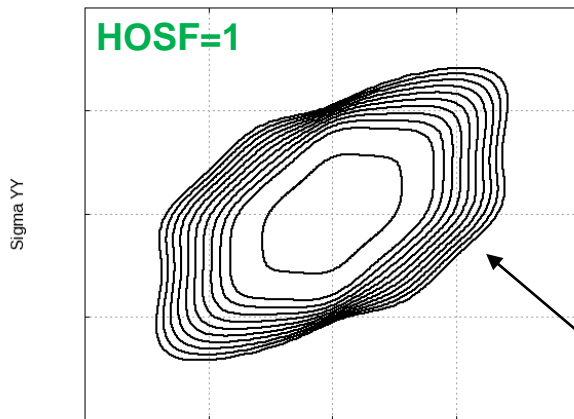
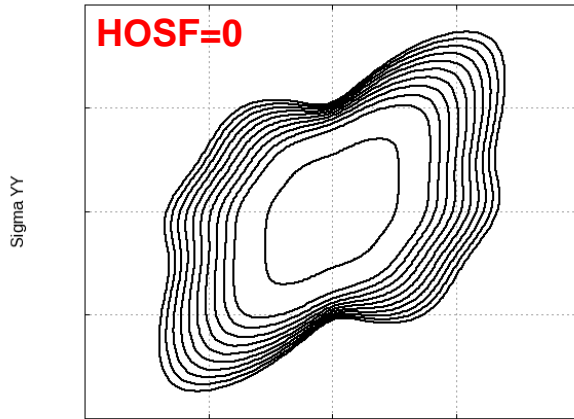
*non-associated plasticity!*



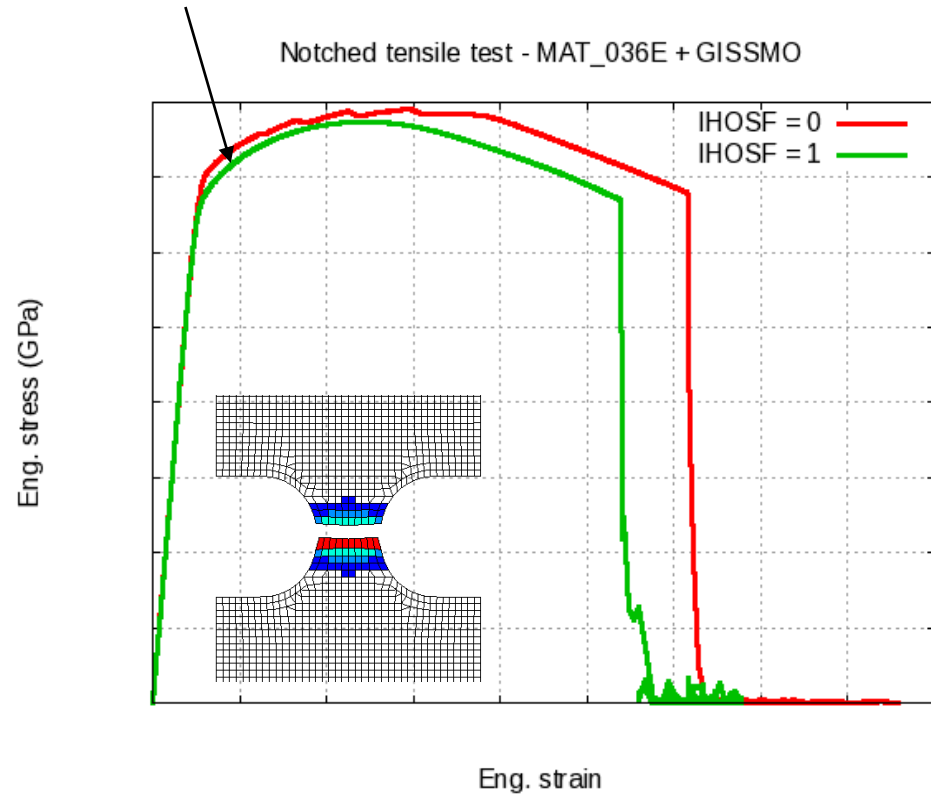


# \*MAT\_036E + HOSF=0/1

Comparison between the formulations (aluminum extrusion)



*No oscillations with HOSF=1*



*Yield surface is much more well-behaved with HOSF=1*



# **EXAMPLE: ALUMINUM SHEET (t=1.5mm)**

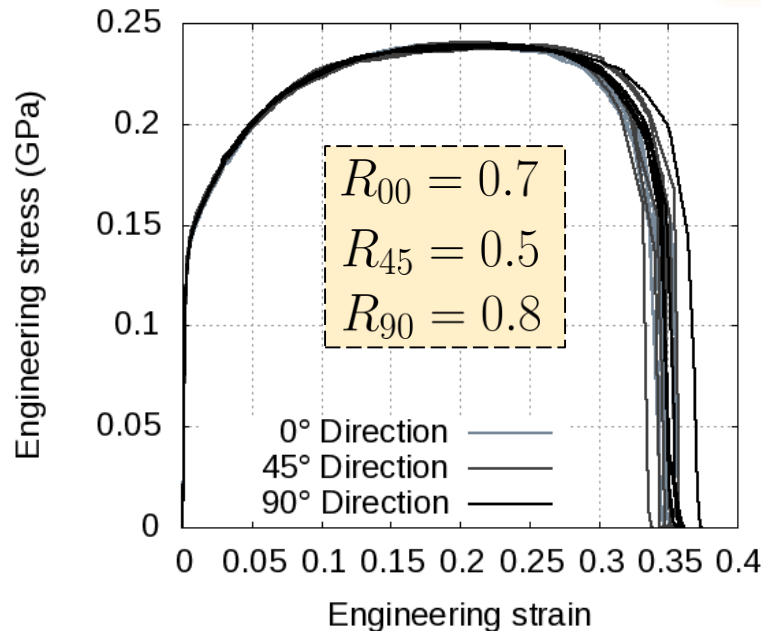
# Aluminum sheet (t=1.5mm)

Experiments performed at DYNAmore in Stuttgart

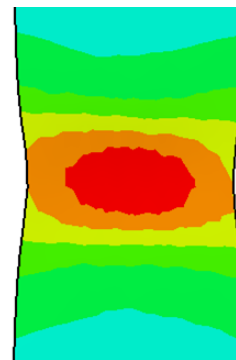
**Workshop:**  
"Material Characterization"  
15<sup>th</sup> Oct 2018



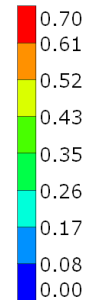
DYNAmore GmbH  
Industriestr. 2  
70565 Stuttgart



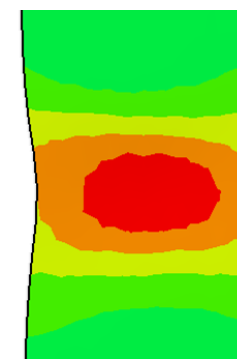
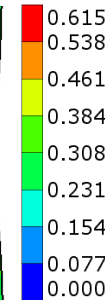
Strain fields prior to failure (DIC measurement)



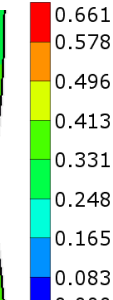
0° Direction



45° Direction



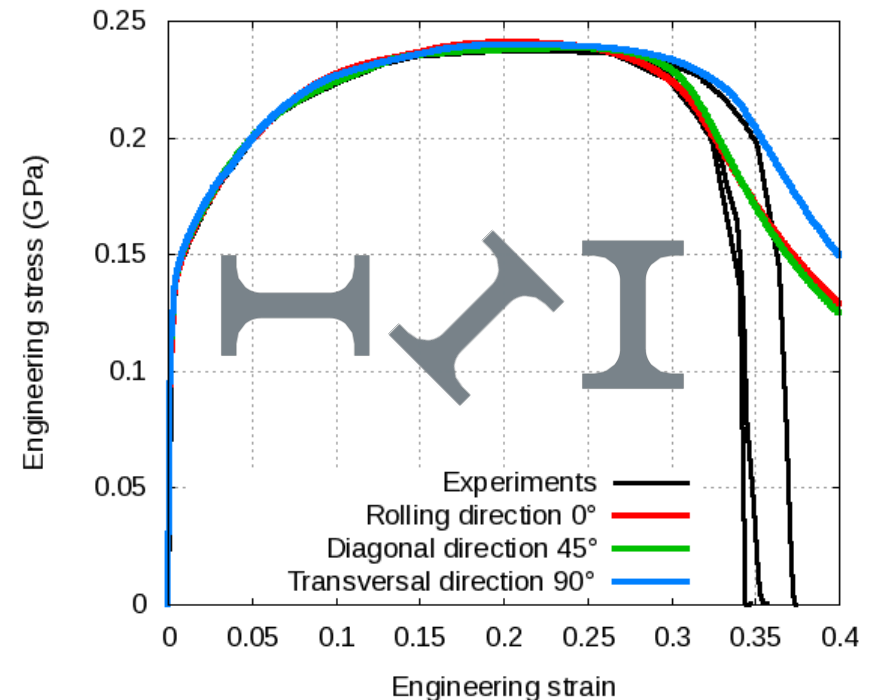
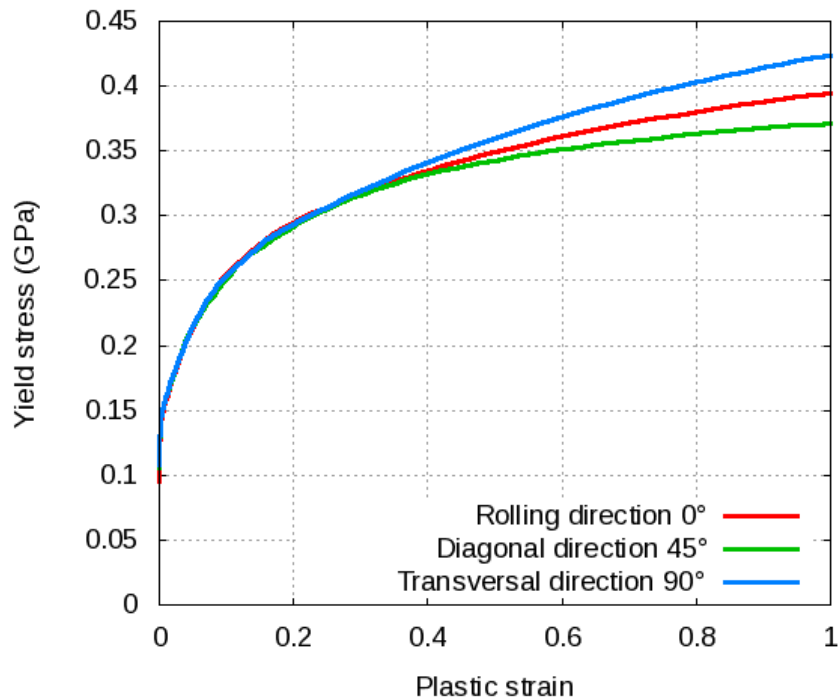
90° Direction



# Hardening curves

Rolling (0°), diagonal (45°) and transversal (90°) directions

- Shell elements of formulation **ELFORM=16**
- Element size for the calibration:  **$L_e=0.5\text{mm}$**
- Material model: **\*MAT\_036E, HOSF=1**
- Constant R values assumed:  **$R_{00}=0.7$ ,  $R_{45}=0.5$ ,  $R_{90}=0.8$**
- Calibration through “reverse engineering”

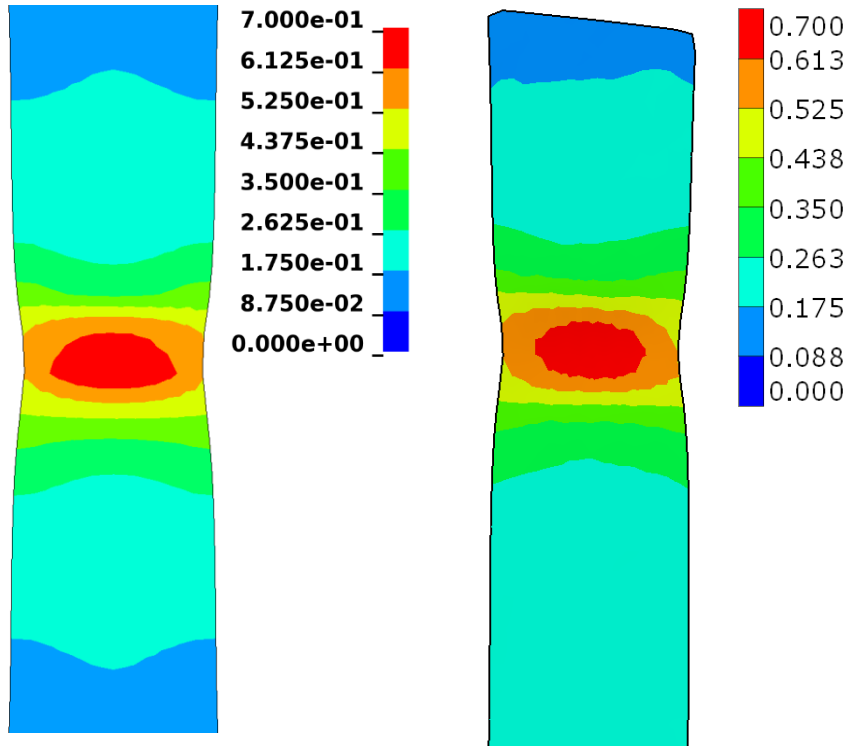


# Small tensile test

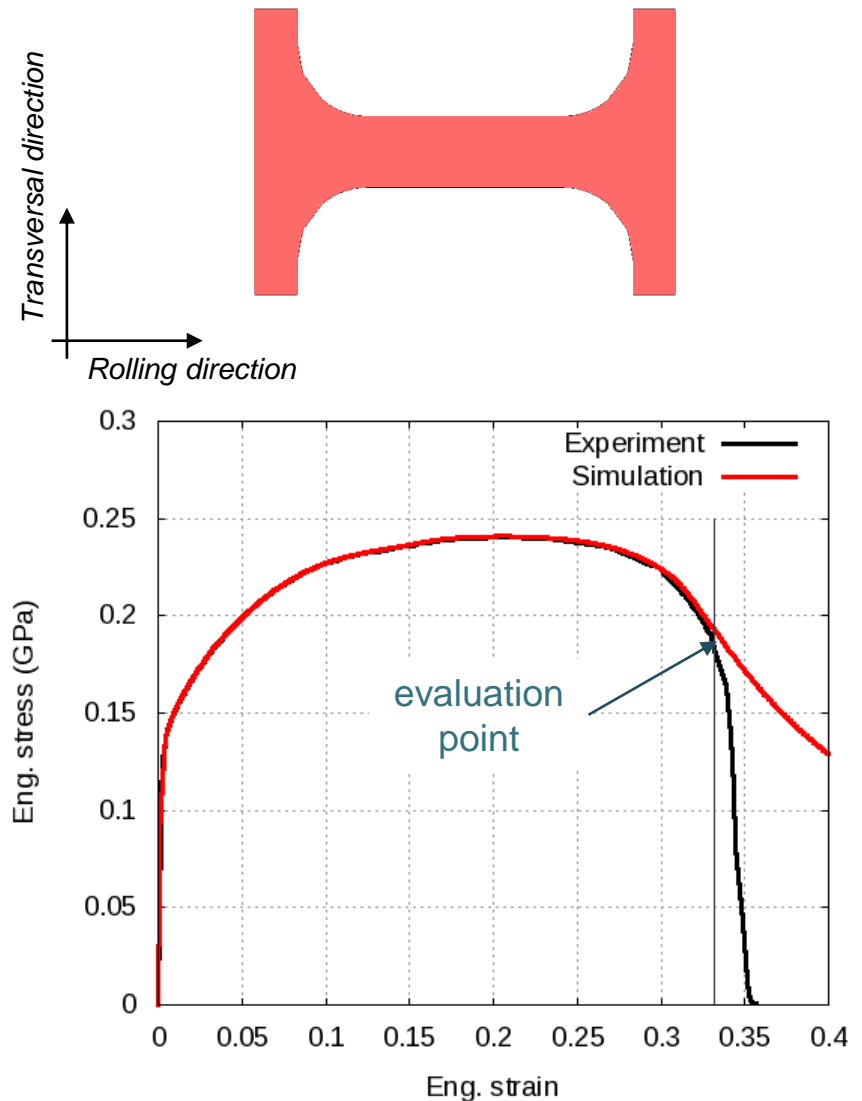
Equivalent strain, **rolling direction (0°)**

LS-DYNA  
(\*MAT\_036E)

Experiment  
(DIC with ARAMIS)



Shell elements, EFORM=16, Le=0.5mm

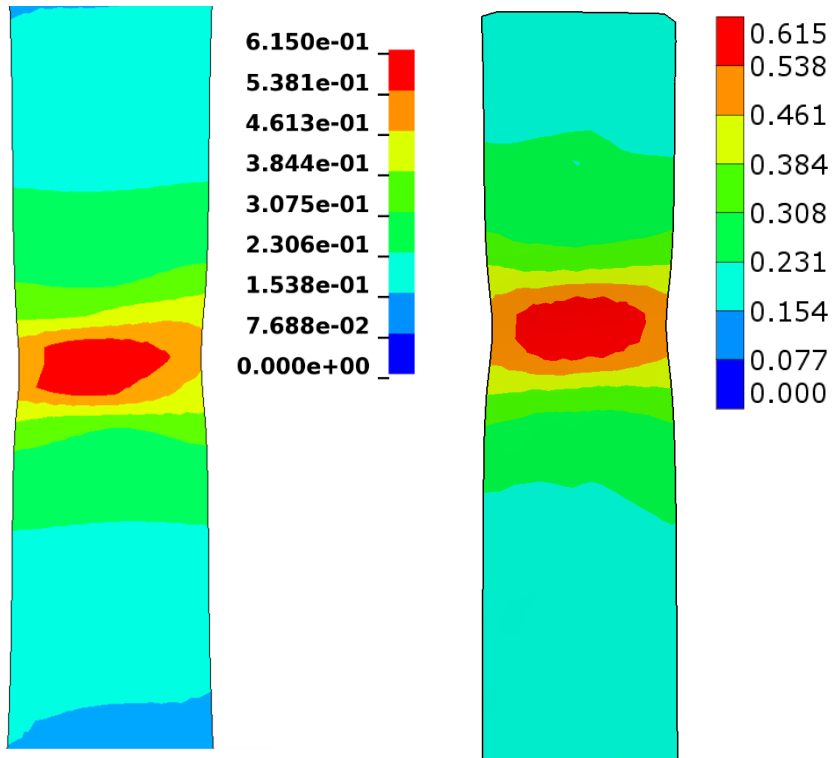


# Small tensile test

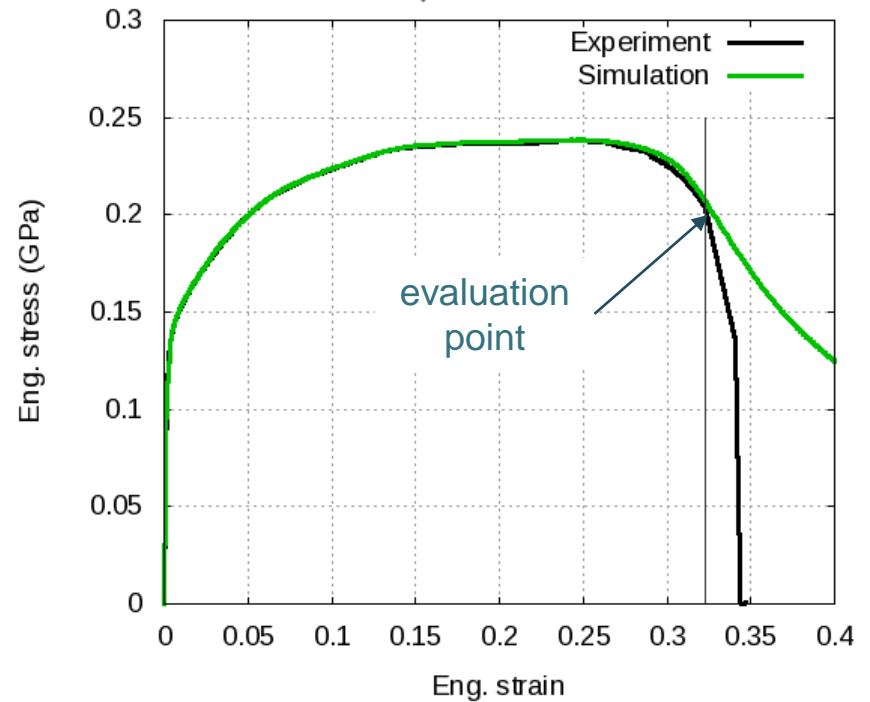
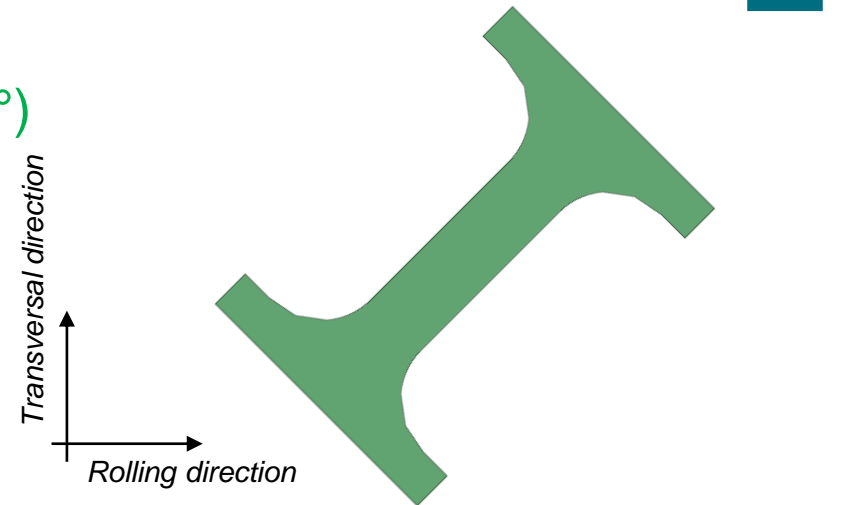
Equivalent strain, diagonal direction (45°)

LS-DYNA  
(\*MAT\_036E)

Experiment  
(DIC with ARAMIS)



Shell elements, EFORM=16, Le=0.5mm

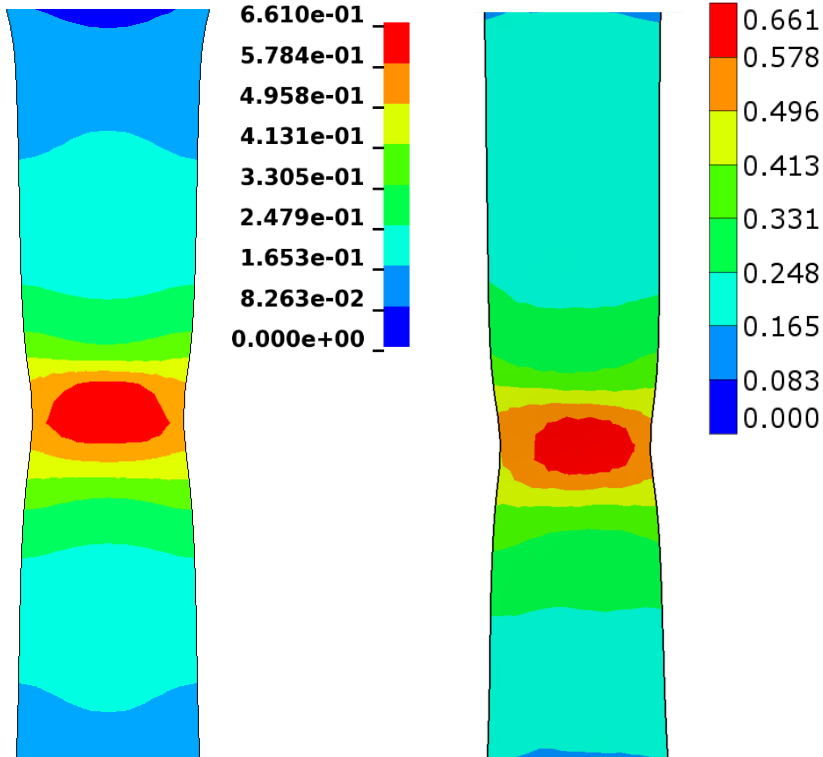


# Small tensile test

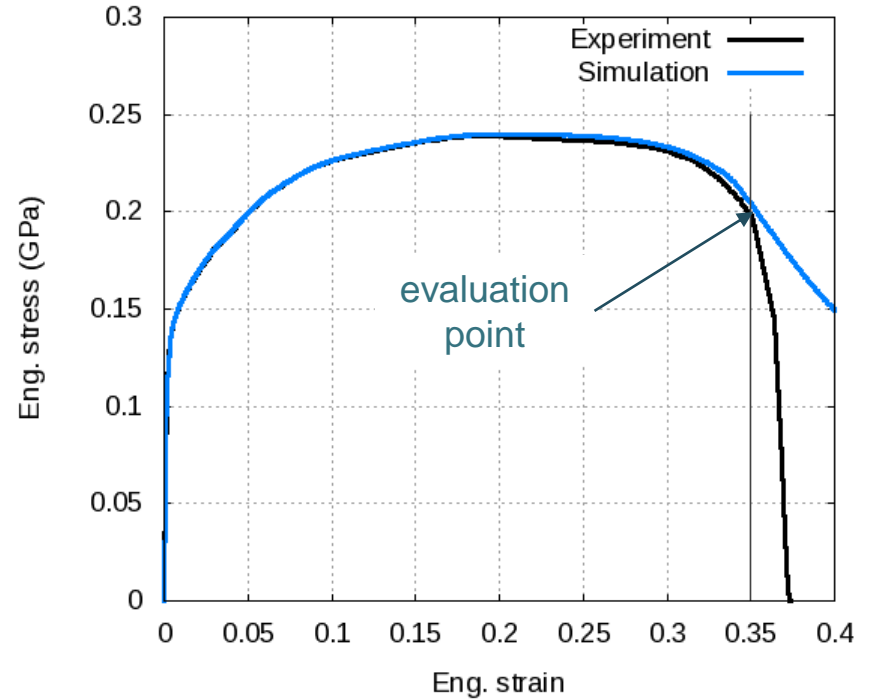
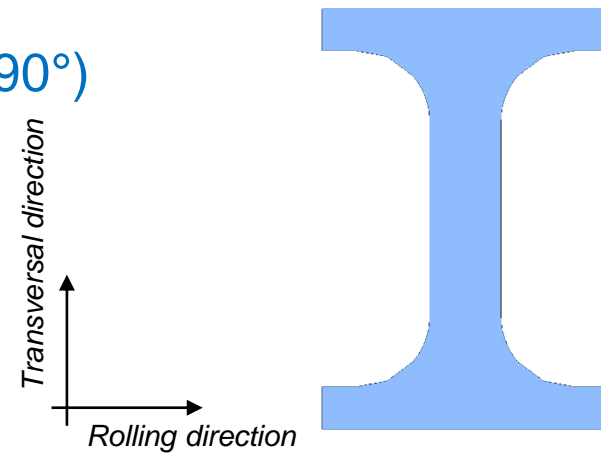
Equivalent strain, transversal direction (90°)

LS-DYNA  
(\*MAT\_036E)

Experiment  
(DIC with ARAMIS)

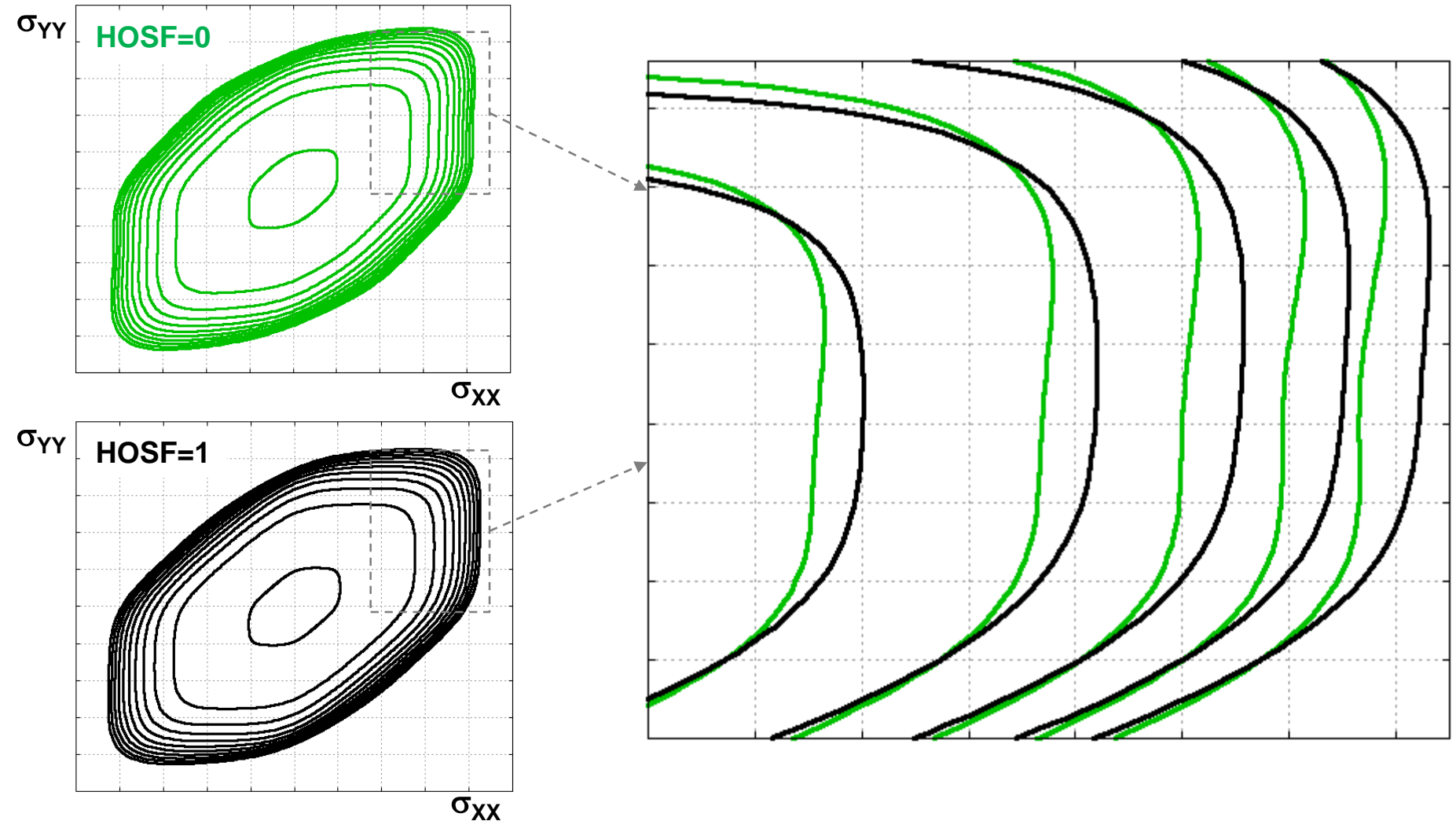


Shell elements, EFORM=16, Le=0.5mm



# Yield Surface (\*MAT\_036E)

Comparison between Barlat-based (HOSF=0) and Hosford-based (HOSF=1)

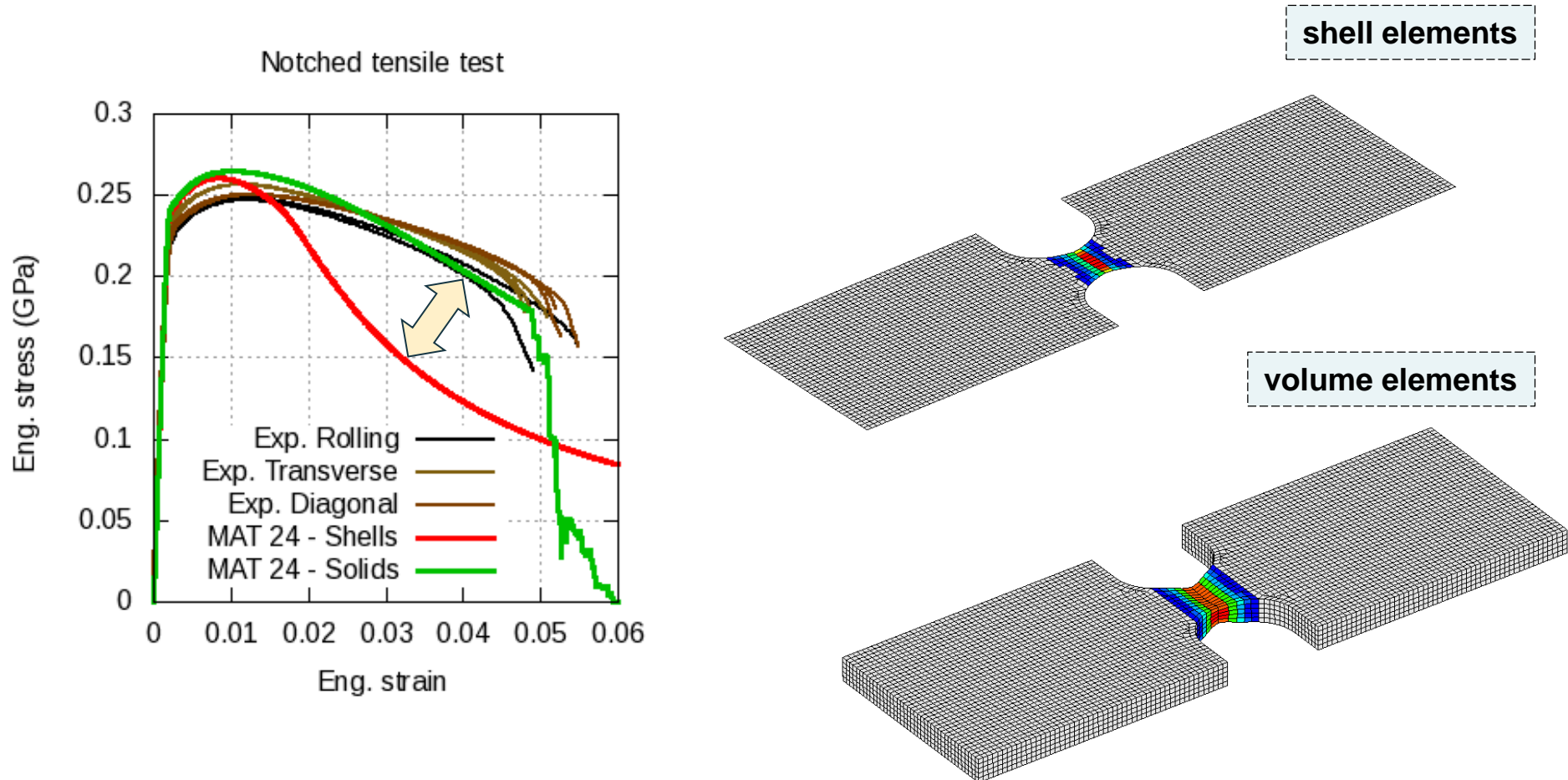




# Modeling with standard shell elements

Limits of the modeling approach

Example: Aluminum extrusion with  $t=2.5\text{mm}$



# Conclusions

## Final remarks and outlook

- Good plasticity is very important for accurate local strains (DIC measurements make validation possible)
- Plastic strain “not only coming from hardening curve”
- \*MAT\_036E is very flexible for matching experimental data (concave yield surfaces still possible, but generally only for “extreme” data)
- Modeling with standard shell elements still poses some barriers...
- Extension to volume elements not quite straightforward...  
... but we're working on that too



# Thank you!



Your LS-DYNA distributor and more